

NASA TT F-9440

FACILITY FORM 602

N65-27720	
(ACCESSION NUMBER)	(THRU)
4	1
(PAGES)	(CODE)
	32 32
(NASA CR OR TMX OR AD NUMBER)	(CATEGORY)

NASA TT F-9440

INFLUENCE OF THE EDGES OF THE TIPS ON THE MOTION OF THE WING
WITH VIBRATIONS AT SUPERSONIC SPEED

Ye. A. Krasil'shchikova

GPO PRICE \$ _____

OTS PRICE(S) \$ _____

Hard copy (HC) 1.00

Microfiche (MF) 50

Translation of "Vliyaniye kontsevykh kromok pri dvizhenii kryla
s vibratsiyami so sverkhzvukovoy skorost'yu."

Doklady AN SSSR, Vol. 58, No. 5,
Aerodinamika, pp. 761-762, 1947

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
WASHINGTON

JULY 1965

INFLUENCE OF THE EDGES OF THE TIPS ON THE MOTION OF THE WING WITH VIBRATIONS AT SUPERSONIC SPEED

Ye. A. Krasil'shchikova

Note 1 showed that to determine $\partial\varphi/\partial z$ in the region EDF and $E_1D_1F_1$ (fig.1) with wing vibration it is /761*

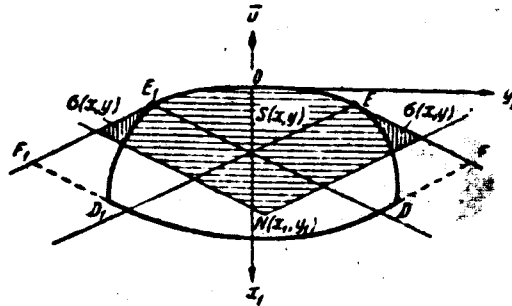


Figure 1

necessary to transform the integral equation

$$\int_0^x \int_{\varphi(\xi)}^y \frac{\Theta(\xi, \eta) \cos \{ \lambda V(x-\xi)(y-\eta) \}}{V(x-\xi)(y-\eta)} d\eta d\xi =$$

$$= - \int_{s(x, \eta)}^x \int_{\varphi(\xi)}^y \frac{A(\xi, \eta) \cos \{ \lambda V(x-\xi)(y-\eta) \}}{V(x-\xi)(y-\eta)} d\eta d\xi \quad (1)$$

relative to function $\Theta(x, y)$.

*Numbers given in the margin indicate the pagination in the original foreign text.

We seek the solution $\theta(1)$ in the form

$$\theta(x, y) = \sum_{n=0}^{\infty} \theta_{2n}(x, y) \lambda^{2n} \quad \left(\lambda = \frac{ad}{u^2 - a^2} \right). \quad (2)$$

In both parts of the equation we add the cosine in the form of a line.

Integrating the lines under the double integral sign, termwise, we reduce (1) to the form

$$\begin{aligned} \sum_{n=0}^{\infty} \lambda^{2n} \sum_{k=0}^n \frac{(-1)^{n-k}}{(2(n-k))!} \int_0^x \int_{\phi(\xi)}^y \theta_{2k}(\xi, \eta) (x-\xi)^{n-k-1/2} (y-\eta)^{n-k-1/2} d\eta d\xi = \\ = \sum_{n=0}^{\infty} \lambda^{2n} \frac{(-1)^{n+1}}{2n!} \int_{s(x,y)} \int A(\xi, \eta) (x-\xi)^{n-1/2} (y-\eta)^{n-1/2} d\eta d\xi. \end{aligned} \quad (3)$$

Equating in (3) the coefficients for single layers λ , we obtain the 762 equations satisfied by $\theta_{2n}(x, y)$:

$$\int_0^x \int_{\phi(\xi)}^y \frac{\theta_{2n}(\xi, \eta)}{\sqrt{(x-\xi)(y-\eta)}} d\eta d\xi = F_n(x, y) \quad (n=0, 1, 2, \dots), \quad (4)$$

where

$$F_n(x, y) = f_n(x, y) + \sum_{k=0}^{n-1} f_n^{(k)}(x, y), \quad (5)$$

in which

$$\begin{aligned} f_n(x, y) &= \frac{(-1)^{n+1}}{2n!} \int_{s(x,y)} \int A(\xi, \eta) (x-\xi)^{n-1/2} (y-\eta)^{n-1/2} d\eta d\xi, \\ f_n^{(k)}(x, y) &= \frac{(-1)^{n-k+1}}{(2(n-k))!} \int_0^x \int_{\phi(\xi)}^y \theta_{2k}(\xi, \eta) (x-\xi)^{n-k-1/2} (y-\eta)^{n-k-1/2} d\eta d\xi, \end{aligned} \quad (6)$$

function $f_n^{(k)}$ determined for $k \geq 0$ and $n > 0$.

For $n = 0$, equation (4) takes the form

$$\int_0^x \int_{\psi(\xi)}^y \frac{\Theta_0(\xi, \eta)}{\sqrt{(x-\xi)(y-\eta)}} d\eta d\xi = f_0(x, y). \quad (7)$$

The solution of this equation is given by formula (7) in note 1.

The function $\Theta_0(x, y)$ corresponds with the values $\partial\varphi/\partial z$ in the case of steady wing motion (for $\lambda = 0$).

The equations in the form (4) for various Θ_{2n} differ from each other only in the form of function $F_n(x, y)$.

If we find the coefficients Θ_{2k} for $k = 0, 1, 2, \dots, n-1$, then $F_n(x, y)$ is the known function in the equation satisfied by $\Theta_{2n}(x, y)$.

We note that for any n , the function $F_n(0, y) = 0$, the solution of (4) is obtained in the same form as that of (7), if in the solution of the last, instead of the function $f_0(x, y) = f(x, y)$ we place the function $F_n(x, y)$.

Then

$$\begin{aligned} \Theta_{2n}(x, y) = & \frac{1}{\pi^2} \frac{1}{\sqrt{y-\psi(x)}} \int_0^x \frac{\partial}{\partial \xi} \left\{ f_n(\xi, \psi(x)) + \sum_{k=0}^{n-1} f_n^{(k)}(\xi, \psi(x)) \right\} \frac{d\xi}{\sqrt{x-\xi}} \\ & + \frac{1}{\pi^2} \int_0^x \int_{\psi(x)}^y \frac{\partial^2}{\partial \xi \partial \eta} \left\{ f_n(\xi, \eta) + \sum_{k=0}^{n-1} f_n^{(k)}(\xi, \eta) \right\} \frac{d\eta d\xi}{\sqrt{(x-\xi)(y-\eta)}}. \end{aligned} \quad (8)$$

Thus, the solution of (1) is presented in the form of an absolutely convergent line (2) for any values of parameter λ .

Submitted 11 May 1947

REFERENCE

1. Krasil'shchikova, Ye. A. DAN, Vol. 58, No. 4, 1947.